

Computational Fluids Mechanics

Part 1: Numerical methods

Part 2: Turbulence models

Part 3: Practice of numerical simulation

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Computational Fluid Mechanics

Part1: Numerical Methods

Objective:

Numerical methods for fluids mechanics and their accuracy

1. Introduction
2. Numerical methods for fluid mechanics (1) ~ Basic equations
3. Numerical methods for fluid mechanics (2) ~Discretizing schemes
4. Numerical methods for fluid mechanics (3) ~Coupling algorism
5. Numerical methods for fluid mechanics (4) ~Additional problems
6. Reliability of numerical simulation

Summary

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Computational Fluid Mechanics

Part1: Numerical Methods (2a)

1. Introduction
2. Numerical methods for fluid mechanics (1) ~governing equations
 - Governing equations of fluid flow
 - Typical solutions of fluid flow
 - Additional models for complex flow phenomena
3. Numerical methods for fluid mechanics (2) ~Discretizing schemes
4. Numerical methods for fluid mechanics (3) ~Coupling algorithm
5. Numerical methods for fluid mechanics (4) ~Additional problems
6. Reliability of numerical simulation

Governing Eqs. of Fluid mechanics

Continuity Eq.
(mass conservation)

$$\frac{D\rho}{Dt} = -\rho I$$

Momentum eq.

$$\rho \frac{Du_i}{Dt} = -\frac{\partial p}{\partial x_i} + \frac{\mu}{3} \frac{\partial I}{\partial x_i} + \mu \nabla^2 u_i + \rho f_i$$

Energy Eq.

$$\rho c_v \frac{DT}{Dt} = -pI + \lambda \nabla^2 T + \Phi + \dot{q}$$

Eq. of state

$$p = \rho RT$$

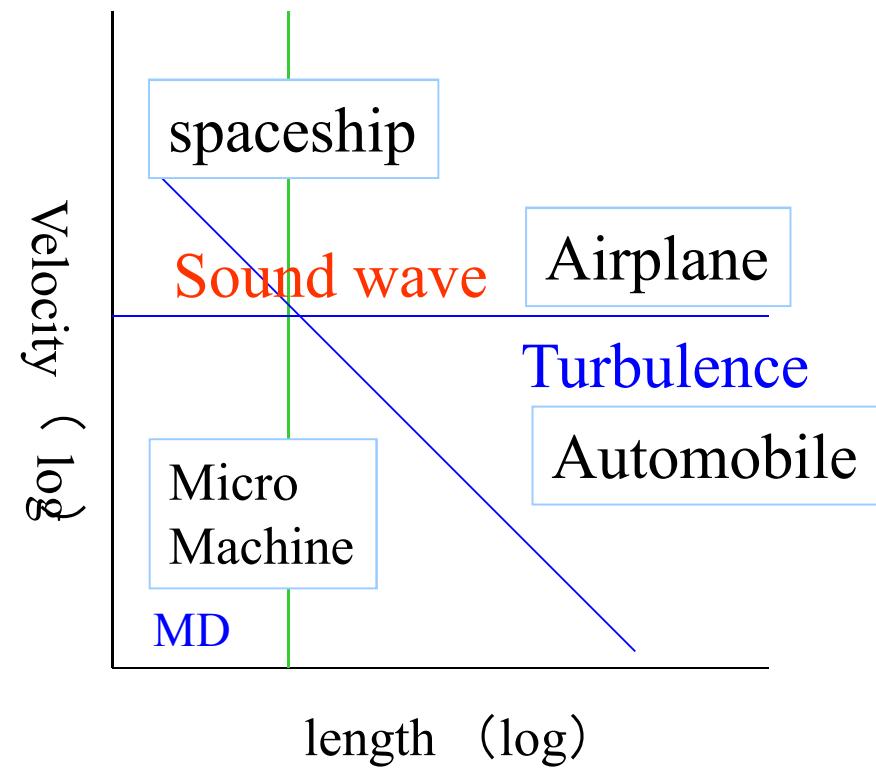
$$; \frac{D}{Dt} \equiv \frac{\partial}{\partial t} + u_j \frac{\partial}{\partial x_j} ; \quad \nabla^2 \equiv \frac{\partial^2}{\partial x_k \partial x_k}$$

$$; \quad I = \frac{\partial u_k}{\partial x_k} \quad ; \quad \Phi = \frac{\partial u_i}{\partial x_j} \left(\frac{\mu}{3} I \delta_{ij} + \mu \frac{\partial u_i}{\partial x_j} \right)$$

Governing Eqs. of Fluid Flow

Non-dimensional Parameters

- **Reynolds no.**
$$\frac{(\text{length}) \times (\text{velocity})}{(\text{viscosity})}$$
- **Mach no.**
$$(\text{velocity}) / (\text{Sound speed})$$
- **Knussen no.**
$$(\text{free pass}) / (\text{length})$$



Non-dimensional Parameters

- in the governing eq.-

$$\left[\frac{L}{t_0 U} \right] \frac{\partial \rho^*}{\partial t^*} + u_j^* \frac{\partial \rho^*}{\partial x_j^*} = -\rho^* \frac{\partial u_j^*}{\partial x_j^*}$$

$$p^* = \rho^* T^*$$

$$\frac{p_0}{\rho_0 T_0} = R$$

$$\left[\frac{L}{t_0 U} \right] \rho^* \frac{\partial u_i^*}{\partial t^*} + \rho^* u_j^* \frac{\partial u_i^*}{\partial x_j^*} = - \left[\frac{p_0}{\rho_0 U^2} \right] \frac{\partial p^*}{\partial x_i^*} + \left[\frac{\mu}{\rho_0 U L} \right] \left(\frac{1}{3} \frac{\partial I^*}{\partial x_i^*} + \frac{\partial^2 u_i^*}{\partial x_j^* \partial x_i^*} \right) + \left[\frac{F}{U^2 L} \right] \rho^* f_i^*$$

$$\begin{aligned} & \left[\frac{L}{t_0 U} \right] \rho^* \frac{\partial T^*}{\partial t^*} + \rho^* u_j^* \frac{\partial T^*}{\partial x_j^*} \\ &= - \left[\frac{p_0}{\rho_0 c_v T_0} \right] p^* I^* + \left[\frac{\lambda}{\rho_0 c_v U L} \right] \frac{\partial^2 T^*}{\partial x_j^* \partial x_i^*} + \left[\frac{\mu}{\rho_0 U L} \right] \left[\frac{U^2}{c_v T_0} \right] \Phi^* + \left[\frac{L Q}{\rho_0 c_v U} \right] \dot{q}^* \end{aligned}$$

$$t = t^* t_0, \quad x_j = x_j^* L, \quad u_i = u_i^* U, \quad p = p^* p_0, \quad T = T^* T_0, \quad \rho = \rho^* \rho_0$$

$$f = F f^*, \quad \dot{q} = Q \dot{q}^*$$

Non-dimensional Parameters

- in the governing eq.-

Parameter of fluid flow

$$\left[\frac{L}{t_0 U} \right] = St \quad (\text{Strouhal no.})$$

$$\left[\frac{p_0}{\rho_0 c_v T_0} \right] = \frac{\rho_0 R T_0}{\rho_0 c_v T_0} = \gamma - 1$$

(γ : specific heat ratio)

$$\left[\frac{p_0}{\rho_0 U^2} \right] = \frac{a_0^2}{\gamma U^2} = \frac{1}{\gamma M^2}$$

$$\left[\frac{\lambda}{\rho_0 c_v U L} \right] = Pe = Re \cdot Pr$$

(M:Mach No)

(Pe:Peclet No., Pr:Prandtl No.)

$$\left[\frac{\mu}{\rho_0 U L} \right] = \frac{1}{Re} \quad (\text{Reynolds No.})$$

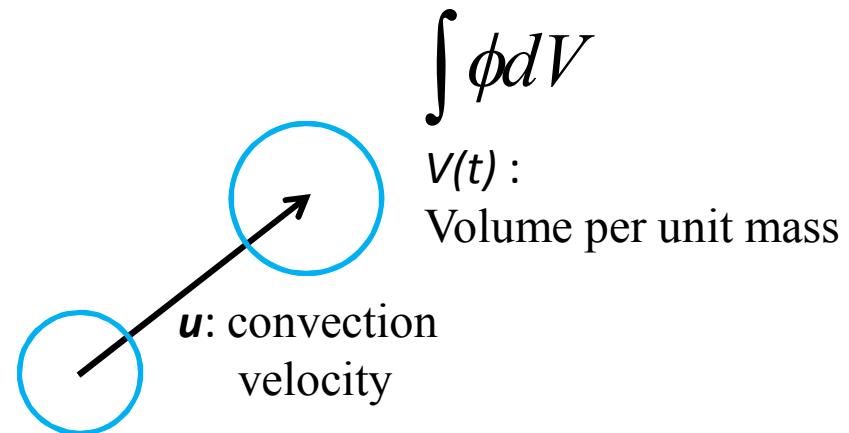
$$\left[\frac{U^2}{c_v T_0} \right] = (\gamma - 1) M^2$$

Lagrangean vs. Eulerian formulations

Lagrangean

$$\frac{D\phi}{Dt} \equiv \left(\frac{\partial \phi}{\partial t} + \mathbf{u} \cdot \nabla \phi \right)$$

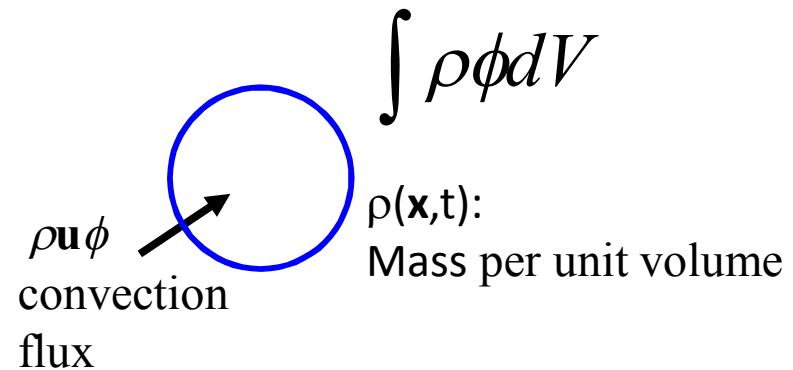
on unit mass [*/kg]



Eulerian

$$\frac{\partial \rho \phi}{\partial t} + \nabla \cdot \rho \mathbf{u} \phi$$

$$\equiv \rho \frac{D\phi}{Dt} + \phi \left(\frac{\partial \rho}{\partial t} + \nabla \cdot \rho \mathbf{u} \right)$$



on unit volume [*/ m^3]

Conservation law

Mass flux vector

$$\frac{\partial \rho}{\partial t} = -\nabla \cdot (\rho \mathbf{u})$$

$$\frac{\partial \rho}{\partial t} = -\left(\frac{\partial \rho u}{\partial x} + \frac{\partial \rho v}{\partial y} \right)$$

$$\int \frac{\partial \rho}{\partial t} dV = - \int (\rho \mathbf{u}) \cdot \mathbf{n} dS$$

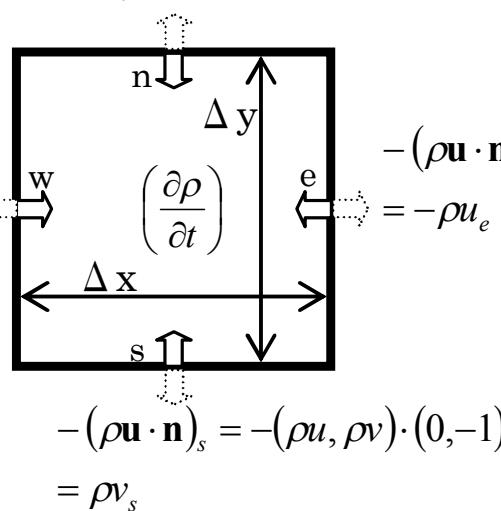
$$\left(\frac{\partial \rho}{\partial t} \right) \Delta x \Delta y \approx (\rho u_w - \rho u_e) \Delta y + (\rho v_s - \rho v_n) \Delta x$$

$$-(\rho \mathbf{u} \cdot \mathbf{n})_n = -(\rho u, \rho v) \cdot (0, 1)$$

$$= -\rho v_n$$

$$-(\rho \mathbf{u} \cdot \mathbf{n})_w =$$

$$-(\rho u, \rho v) \cdot (-1, 0) = \rho u_w$$



Conservation law

Flux vector vs. Stress tensor

$$\frac{\partial \rho \mathbf{u}^t}{\partial t} = -\nabla \cdot (\rho \mathbf{u} \mathbf{u}^t) - \nabla \cdot \mathbf{T} - \nabla \cdot \mathbf{P} + \mathbf{f}^t \quad \Rightarrow \quad \frac{\partial \rho u}{\partial t} = -\nabla \cdot (\rho \mathbf{u} u) - \nabla \cdot \boldsymbol{\tau}_x - \nabla \cdot \mathbf{p}_x + f_x$$

$$\rho \mathbf{u}^t = [\rho u \quad \rho v], \quad \mathbf{f}^t = [f_x \quad f_x] \quad \rho \mathbf{u} \mathbf{u}^t = [\rho \mathbf{u} u \quad \rho \mathbf{u} v] = \begin{bmatrix} \rho u u & \rho u v \\ \rho v u & \rho v v \end{bmatrix},$$

$$\mathbf{P} = [\mathbf{p}_x \quad \mathbf{p}_y] = \begin{bmatrix} p & 0 \\ 0 & p \end{bmatrix} (= p \mathbf{I} = p \delta_{ij}) \quad \therefore \nabla \cdot \mathbf{P} = (\nabla p)^t = \begin{bmatrix} \frac{\partial p}{\partial x} & \frac{\partial p}{\partial y} \end{bmatrix}$$

$$\mathbf{T} = [\boldsymbol{\tau}_x \quad \boldsymbol{\tau}_y] = \begin{bmatrix} -2\mu \frac{\partial u}{\partial x} & -\mu \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \\ -\mu \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) & -2\mu \frac{\partial v}{\partial y} \end{bmatrix} (= 2\mu \mathbf{S})$$

Conservation law

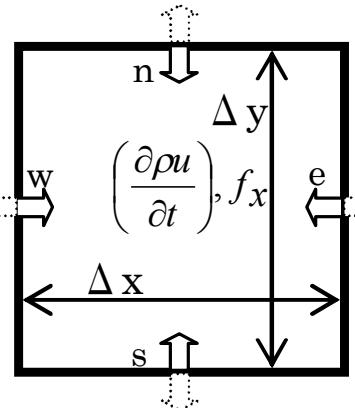
Momentum flux vector

$$\frac{\partial \rho u}{\partial t} = -\nabla \cdot (\mathbf{J}_{conv.} + \mathbf{J}_{diff.} + \mathbf{J}_{pres.}) + f_x$$

$$\mathbf{J}_{conv.} = \rho \mathbf{u} u = \begin{bmatrix} \rho u u \\ \rho v u \end{bmatrix}, \quad \mathbf{J}_{diff.} = \begin{bmatrix} -2\mu(\partial u / \partial x) \\ -\mu(\partial u / \partial y + \partial v / \partial x) \end{bmatrix}, \quad \mathbf{J}_{pres.} = \begin{bmatrix} p \\ 0 \end{bmatrix}$$

$$\int \frac{\partial \rho u}{\partial t} dV = - \int (\mathbf{J}_{conv.} + \mathbf{J}_{diff.} + \mathbf{J}_{pres.}) \cdot \mathbf{n} dS + \int f_x dV$$

$$\begin{aligned} -(\mathbf{J}_{conv.} \cdot \mathbf{n})_w &= J_{conv.} = (\rho uu)_w \\ -(\mathbf{J}_{diff.} \cdot \mathbf{n})_w &= J_{diff.} = -2\mu \frac{\partial u}{\partial x}_w \\ -(\mathbf{J}_{pres.} \cdot \mathbf{n})_w &= J_{pres.} = p_w \end{aligned}$$



$$\begin{aligned} -(\mathbf{J}_{conv.} \cdot \mathbf{n})_e &= -J_{conv.} = -(\rho uu)_e \\ -(\mathbf{J}_{diff.} \cdot \mathbf{n})_e &= -J_{diff.} = 2\mu \frac{\partial u}{\partial x}_e \\ -(\mathbf{J}_{pres.} \cdot \mathbf{n})_e &= -J_{pres.} = -p_e \end{aligned}$$

Exercise 2a

Calculate each component of momentum flux through the vertical surface (n & s in fig.) of considering volume.

Derive the conservation law for y-direction component of momentum (ρv)

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Typical solutions of fluid flow

convection

$$\frac{\partial \rho}{\partial t} = -u_j \frac{\partial \rho}{\partial x_j}$$

$$\rho \frac{\partial u_i}{\partial t} = -\rho u_j \frac{\partial u_i}{\partial x_j}$$

$$\rho c_v \frac{\partial T}{\partial t} = -\rho c_v u_j \frac{\partial T}{\partial x_j}$$

interaction (volume source)

$$-\rho \frac{\partial u_k}{\partial x_k}$$

$$-\frac{\partial p}{\partial x_i} + F_i$$

$$-p \frac{\partial u_k}{\partial x_k} + Q + \Phi$$

$$p = \rho R T$$

diffusion (surface flux)

$$+\frac{\partial}{\partial x_j} \left(\frac{\mu}{3} \frac{\partial u_k}{\partial x_k} \delta_{ij} + 2\mu S_{ij} \right)$$

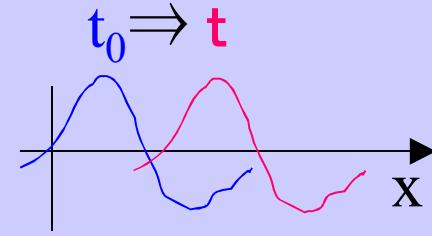
$$+\frac{\partial}{\partial x_j} \left(k \frac{\partial T}{\partial x_j} \right)$$

Solution of convection

- Model of scalar convection

(linear eq.)

$$\frac{\partial \phi}{\partial t} = -u \frac{\partial \phi}{\partial x} \Rightarrow \phi(x, t) = \phi_0(x - ut)$$

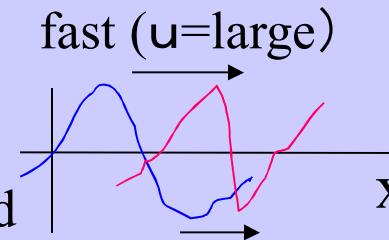


- Model of momentum convection

(Burger's Eq.)

$$\frac{\partial u}{\partial t} = -u \frac{\partial u}{\partial x}$$

- shape deformed
- high frequency produced



- keep linear solution
- adaptive to steep variation

Solution of diffusion and external force

Diffusion of scalar (Linear)

time dependent heat transfer problem

$$\frac{\partial \phi}{\partial t} = \Gamma \frac{\partial^2 \phi}{\partial x^2} \quad \Rightarrow \quad \phi = \phi_n \exp\left(inx - \frac{t}{\Gamma}\right)$$

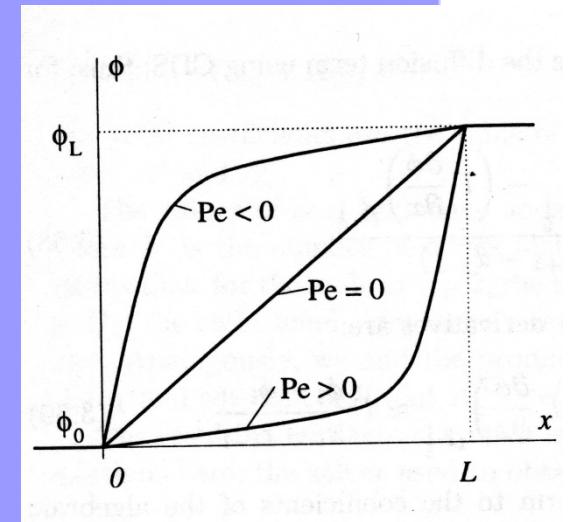
steady convective – diffusion problem

$$u \frac{\partial \phi}{\partial x} = \Gamma \frac{\partial^2 \phi}{\partial x^2} \quad \Rightarrow \quad \left(u\phi - \Gamma \frac{\partial \phi}{\partial x}\right) = \text{const}$$

Peclet No.: $\text{Pe} = uL/\Gamma$

Source term (Linear)

$$\frac{\partial \phi}{\partial t} = -S\phi \quad \Rightarrow \quad \phi = \phi_0 \exp(-St)$$



Solution of Interaction step

Compressible (sound wave)

$$\frac{\partial \rho}{\partial t} = -\rho \frac{\partial u}{\partial x}$$

$$\rho \frac{\partial u}{\partial t} = -\frac{\partial p}{\partial x}$$

$$\frac{p}{\rho^\kappa} = \text{const}$$

(S=const.)

$$\left. \begin{aligned} \frac{\partial \rho}{\partial t} &= -\rho \frac{\partial u}{\partial x} \\ \rho \frac{\partial u}{\partial t} &= -a^2 \frac{\partial \rho}{\partial x} \\ a^2 &\equiv \left(\frac{dp}{d\rho} \right)_s = \sqrt{\kappa \frac{p}{\rho}} \end{aligned} \right\}$$

Mach No. : $\text{Ma} = U / a$
a: sound speed
U: flow velocity

$$\frac{\partial^2 u}{\partial t^2} = a^2 \frac{\partial^2 u}{\partial x^2} \quad \text{Sound wave}$$

$$\Rightarrow u = f(x - at) + g(x + at)$$

Solution of Interaction step

Incompressible flow

$$\nabla \cdot \mathbf{u} = \frac{\partial u_i}{\partial x_i} = 0$$

$$\frac{\partial u_i}{\partial t} = -\frac{1}{\rho} \frac{\partial p}{\partial x_i} + F_i$$

$$(F_i = -u_j \frac{\partial u_i}{\partial x_j} + \dots)$$

$$\rho = \text{const}$$

if $\text{Ma} \rightarrow 0$
 $a(\text{sound speed}) \rightarrow \infty$

$$\frac{1}{\rho} \nabla \cdot \nabla p = -\frac{\partial}{\partial t} (\nabla \cdot \mathbf{u}) + \nabla \cdot \mathbf{F}$$

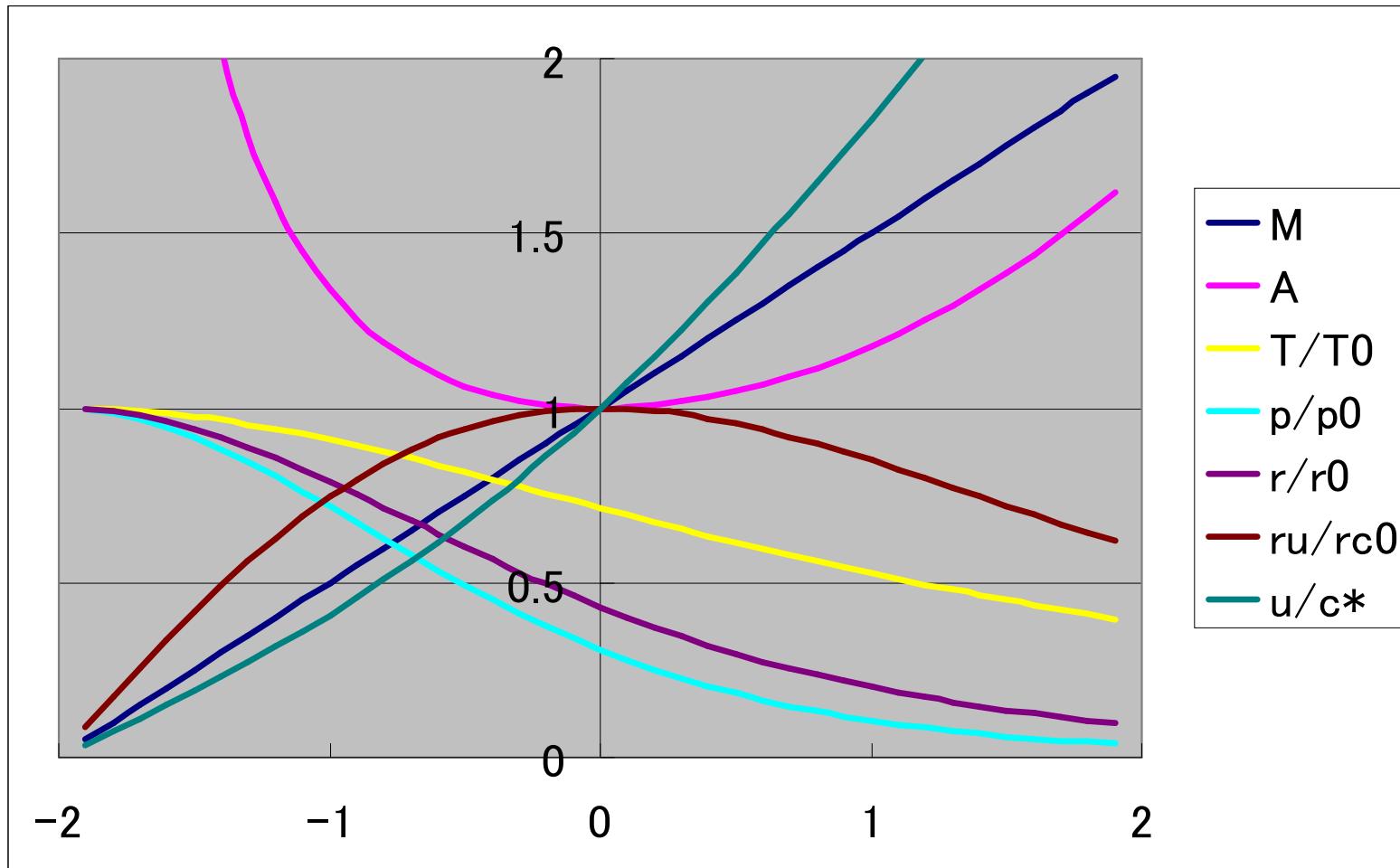
Pressure poison eq.

Solution of Interaction step

Compressible (1D nozzle)

$$\left. \begin{array}{l} \frac{d\rho}{\rho} + \frac{du}{u} + \frac{dA}{A} = 0 \\ u du + \frac{dp}{\rho} = 0 \\ \frac{dp}{p} - \kappa \frac{d\rho}{\rho} = 0 \\ \frac{dp}{p} = \frac{d\rho}{\rho} + \frac{dT}{T} \end{array} \right\} \quad \left. \begin{array}{l} \frac{d\rho}{\rho} = - \frac{M^2}{M^2 - 1} \frac{dA}{A} \\ \frac{du}{u} = + \frac{1}{M^2 - 1} \frac{dA}{A} \\ \frac{dp}{p} = - \frac{\kappa M^2}{M^2 - 1} \frac{dA}{A} \\ \frac{dT}{T} = - \frac{(\kappa - 1)M^2}{M^2 - 1} \frac{dA}{A} \end{array} \right\} \quad \left. \begin{array}{l} \frac{da}{a} = - \frac{(\kappa - 1)M^2}{2(M^2 - 1)} \frac{dA}{A} \\ \frac{dM}{M} = + \frac{2 + (\kappa - 1)M^2}{M^2 - 1} \frac{dA}{A} \end{array} \right.$$

Solution of 1D compressible flow



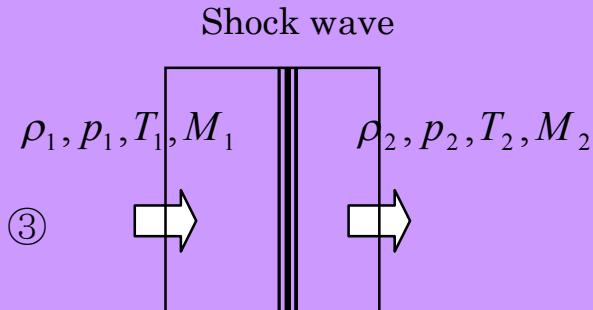
Solution of Interaction step

Compressible (1D shock wave)

$$\text{Mass} : \rho_1 u_1 = \rho_2 u_2 \quad \textcircled{1}$$

$$\text{Momentum} : p_1 + \rho_1 u_1^2 = p_2 + \rho_2 u_2^2 \quad \textcircled{2}$$

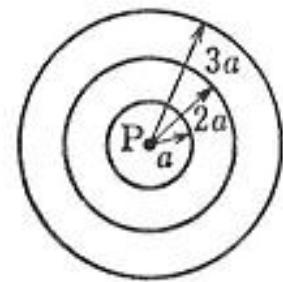
$$\text{Energy} : \frac{\kappa}{\kappa-1} \frac{p_1}{\rho_1} + \frac{1}{2} u_1^2 = \frac{\kappa}{\kappa-1} \frac{p_2}{\rho_2} + \frac{1}{2} u_2^2 \quad \textcircled{3}$$



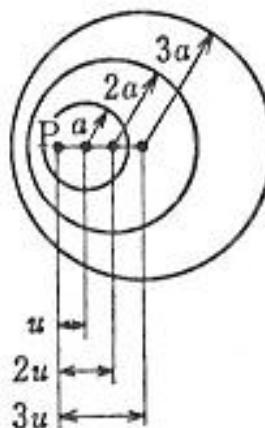
$$\therefore \frac{\rho_2}{\rho_1} = \frac{\left(\frac{\kappa+1}{\kappa-1} \frac{p_2}{p_1} + 1 \right)}{\left(\frac{p_2}{p_1} + \frac{\kappa+1}{\kappa-1} \right)} = \frac{u_1}{u_2} > 1 \quad \text{Rankine-Hugoniot eq.}$$

$$\therefore \frac{T_2}{T_1} = \frac{p_2}{p_1} \frac{\rho_1}{\rho_2} = \frac{p_2 \left(\frac{\kappa+1}{\kappa-1} \frac{p_2}{p_1} + 1 \right)}{\left(\frac{p_2}{p_1} + \frac{\kappa+1}{\kappa-1} \right)} = \frac{\left(\frac{p_2}{p_1} + \frac{\kappa+1}{\kappa-1} \right)}{\left(\frac{p_1}{p_2} + \frac{\kappa+1}{\kappa-1} \right)} > 1$$

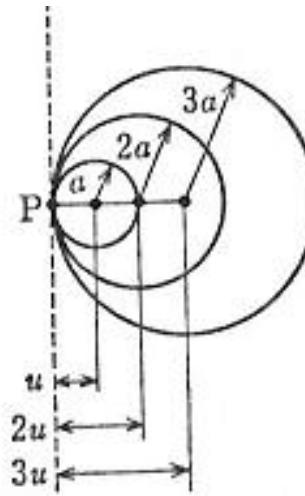
Profiles of shock wave



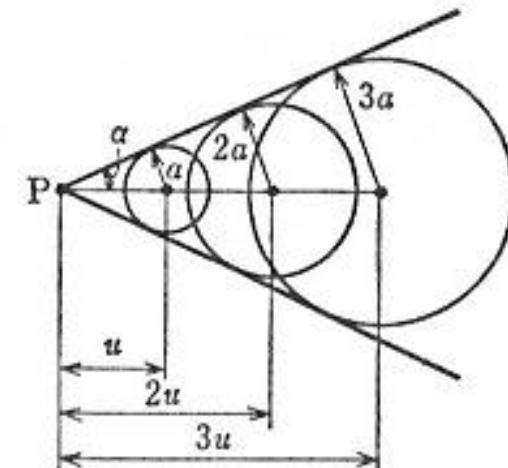
(a) $u=0, M=0$



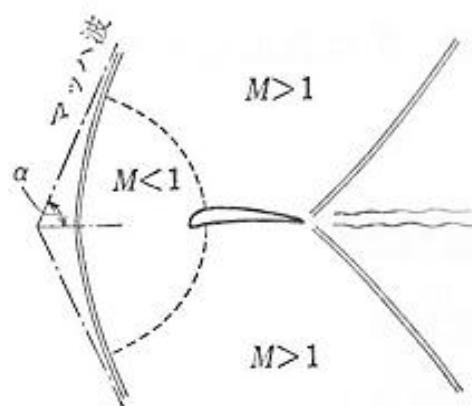
(b) $u < a, 0 < M < 1$



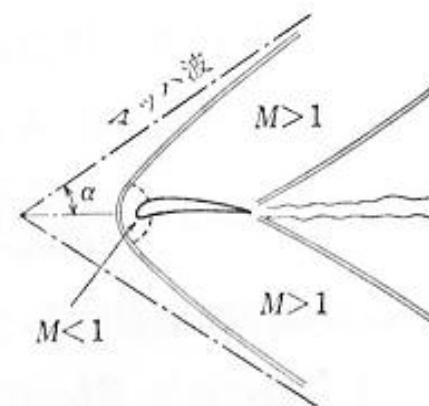
(c) $u=a, M=1$



(d) $u > a, M > 1$



(d) M_∞ medium



(e) M_∞ large

Exercise 2b

Solve the following equations of $\phi(x)$ and draw the profiles for different parameter values.

$$u \frac{\partial \phi}{\partial x} = \Gamma \frac{\partial^2 \phi}{\partial x^2} \quad u, \Gamma \text{ are constant parameter}$$

$$0 \leq x \leq L \quad \phi = 0 \quad \text{at } x = 0$$

$$\phi = 1 \quad \text{at } x = L$$

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Complexity of system

Strong coupling

convection	interaction	diffusion	external-force	
$\frac{\partial \rho}{\partial t}$	$-u_j \frac{\partial \rho}{\partial x_j}$	$-\rho \frac{\partial u_k}{\partial x_k}$		+ M
$\rho \frac{\partial u_i}{\partial t}$	$-\rho u_j \frac{\partial u_j}{\partial x_i}$	$-\frac{\partial p}{\partial x_i}$		mass
$\rho c_v \frac{\partial T}{\partial t}$	$-\rho c_v u_j \frac{\partial T}{\partial x_j}$	$-p \frac{\partial u_k}{\partial x_k}$	$+\frac{\partial}{\partial x_j} \left(\frac{\mu}{3} I \delta_{ij} + 2\mu S_{ij} \right) + F_i$	momentum
		$p = \rho R T$	$+\frac{\partial}{\partial x_j} \left(k \frac{\partial T}{\partial x_j} \right)$	energy

- 1) **hyperbolic(compressible)**
- 2) **harmonic(sound wave)**
- 3) **parabolic(incompressible)**

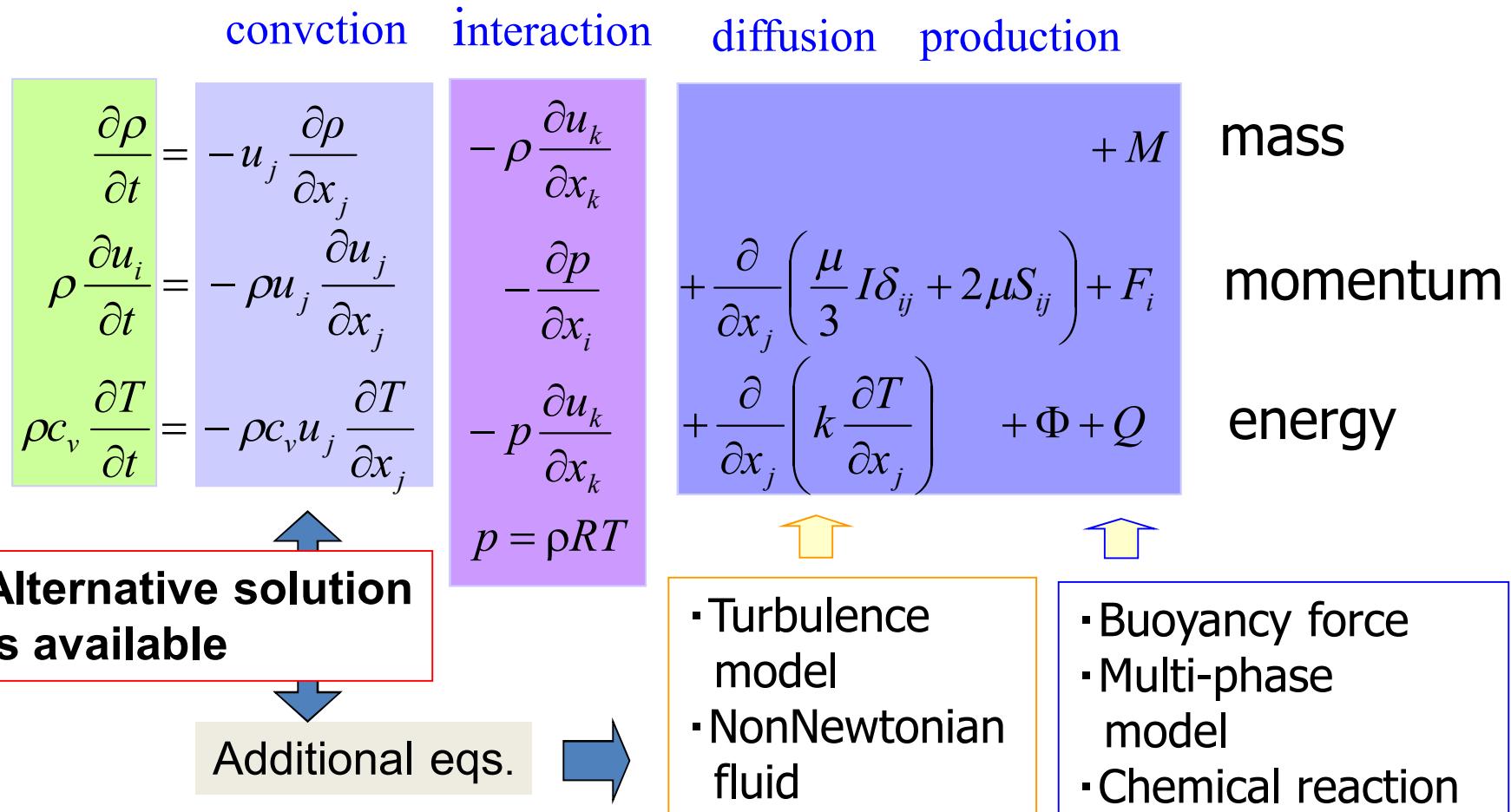
Ma>1

Ma<1

Combined solution
Is needed

Complexity of system

Weak coupling



Model of Combustion Flow

– Basic eqs. of reactive flow –

Mass:

$$\frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x_i} (\rho u_j) = 0$$

Momentum:

$$\frac{\partial \rho u_i}{\partial t} + \frac{\partial}{\partial x_j} (\rho u_i u_j) = - \frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} (2 \mu S_{ij}) + \frac{\partial}{\partial x_j} \left\{ \left(\mu_V - \frac{2}{3} \mu \right) \frac{\partial u_k}{\partial x_k} \delta_{ij} \right\}$$

Energy:

$$\rho C_p \frac{\partial T}{\partial t} + \rho C_p u_j \frac{\partial T}{\partial x_j} = \frac{\partial}{\partial x_j} \left[\frac{\mu}{Pr} \frac{\partial T}{\partial x_j} \right] + \frac{\partial p}{\partial t}$$

Eq.of gas state: $p = \rho R_o T \sum_{\alpha=1}^N Y_{\alpha} / M_{\alpha}$

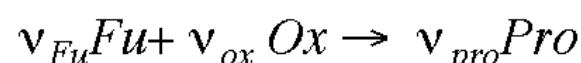
$$+ \left[\sum_{\alpha=1}^N C_p \frac{\mu}{Sc_{\alpha}} \frac{\partial Y_{\alpha}}{\partial x_j} \right] \frac{\partial T}{\partial x_j} - \sum_{\alpha=1}^N h_{0\alpha} \omega_{\alpha}$$

+

Chemical species $\frac{\partial \rho Y_{\alpha}}{\partial t} + \frac{\partial}{\partial x_j} (\rho Y_{\alpha} u_j) = \frac{\partial}{\partial x_j} \left[\frac{\mu}{Sc_{\alpha}} \frac{\partial Y_{\alpha}}{\partial x_j} \right] + \omega_{\alpha}$

Arrhenius' law

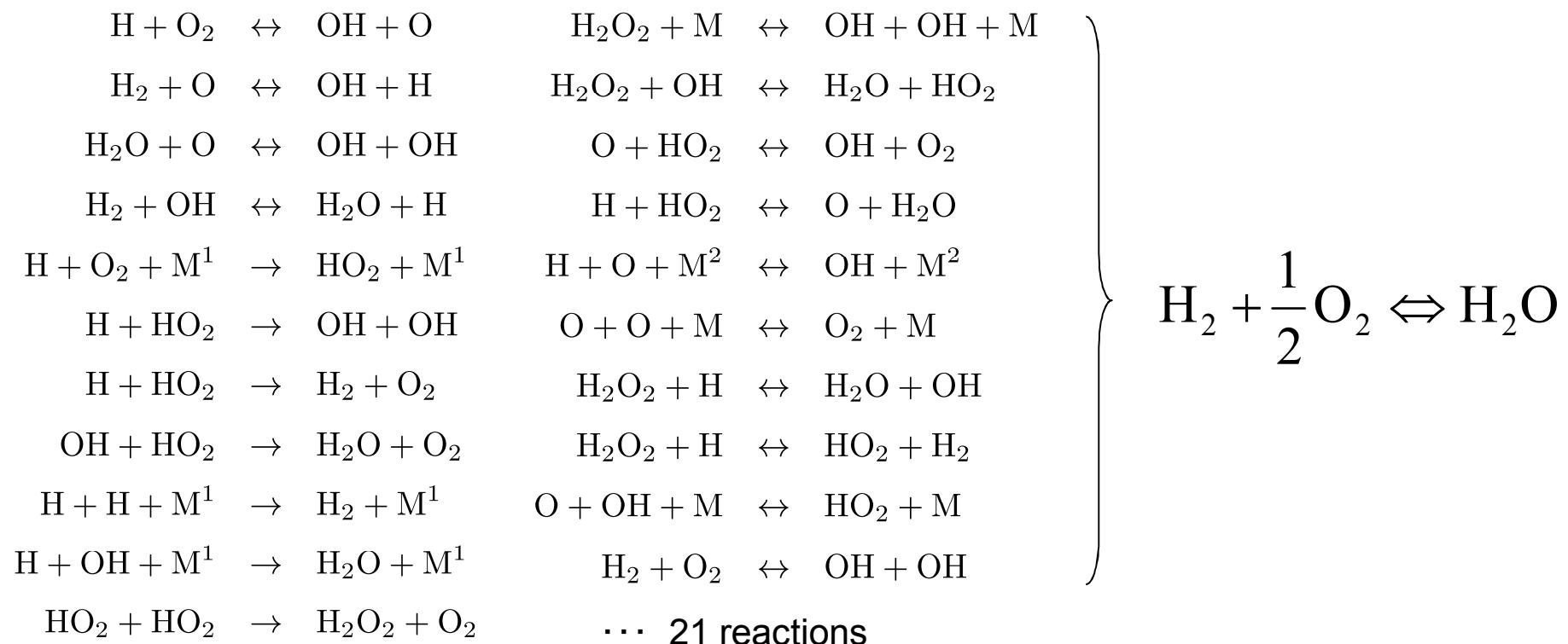
Chemical reaction



$$\omega_{fu} = -B \exp \left(-\frac{E}{RT} \right) \left(\frac{\rho Y_{fu}}{M_{fu}} \right)^{v_{fu}} \left(\frac{\rho Y_{ox}}{M_{ox}} \right)^{v_{ox}}$$

Detail reactions of H₂·O₂ flame

Spices	N ₂	O ₂	H ₂	H ₂ O	OH
Lewis No.	1	1.11	0.3	1.12	0.73
Spices	O	H	H ₂ O ₂	HO ₂	-
Lewis No.	0.7	0.18	1.12	1.1	-



Model of Combustion Flow

– non-dimensional parameters of mass & heat transfer –

- Nusselt no. $Nu \equiv \frac{h\lambda}{L}$ h: heat transfer coefficient
λ: Thermal conductivity
 - Prandtl no. $Pr \equiv \frac{\nu}{\alpha}$ ν: kinetic viscosity
 - Schmidt no. $Sc \equiv \frac{\nu}{D}$ D: mass diffusion conductivity
 - Lewis no. $Le \equiv \frac{\alpha}{D}$ α: temp. diffusion conductivity

Model of Combustion Flow

– non-dimensional parameters of reactive flow –

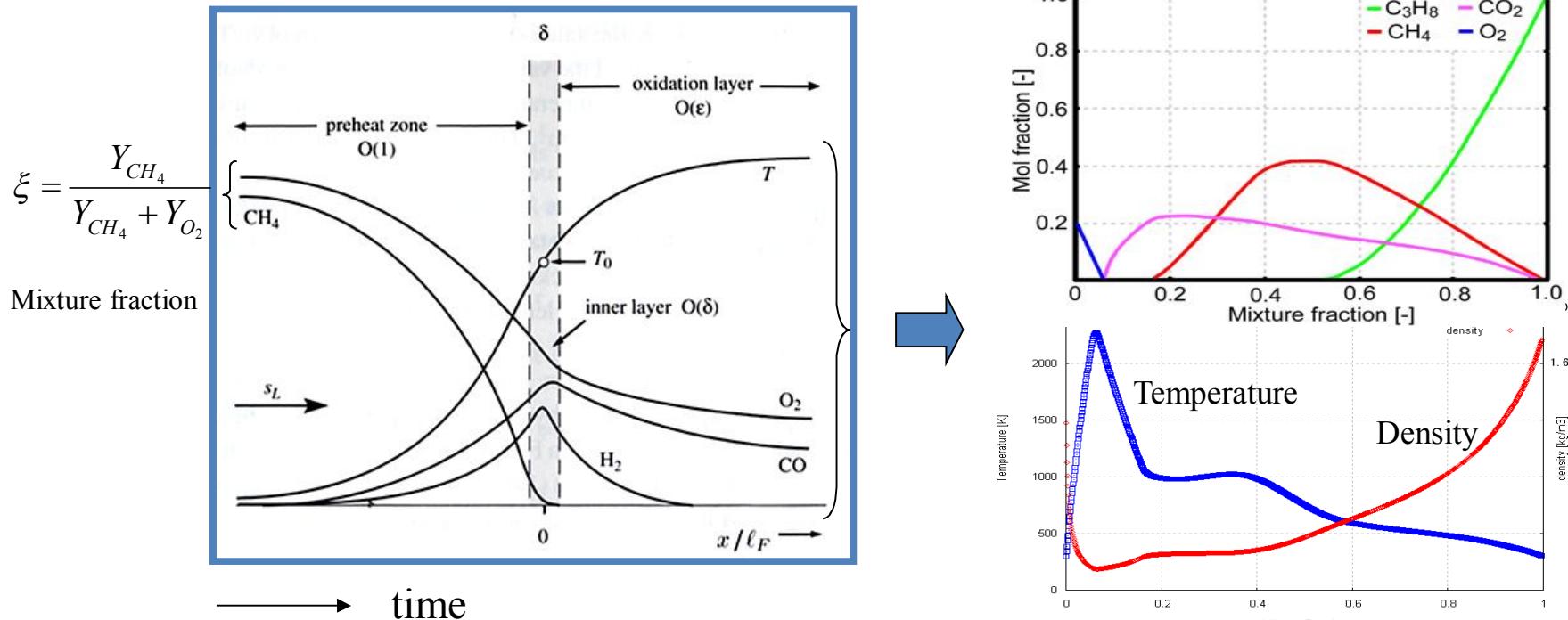
- (primal) Damköhler no. $D \equiv \frac{\tau_r}{\tau_c}$: time scale of flow
 τ_r : time scale of reaction
- Karlovits no. $K \equiv \frac{\eta}{U} \frac{dU}{dy}$ Thickness of heating zone: $\eta = \frac{\lambda}{\rho c_p S} = \frac{\alpha}{S}$
Combustion speed : S
- turbulent Damköhler no.

$$Da \equiv \frac{\tau_t}{\tau_c} = \frac{\ell}{u' \delta}$$

ℓ : turbulent length scale
 u' : velocity fluctuation, δ : flame thickness

Model of Combustion

- mechanics of reactive flow -



Enthalpy: $H = E + PV \Rightarrow dH = (TdS - PdV) + d(PV) = TdS + VdP$ $\left(\frac{\partial H}{\partial S} \right)_p = T > 0$

Free energy: $G = H - TS \Rightarrow dG = (TdS + VdP) - d(TS) = -SdT + VdP$ $\left(\frac{\partial G}{\partial T} \right)_p = -S < 0$

Enthalpy components:

$$dH = \sum_{\alpha} h_{\alpha}^0 dn_{\alpha} + c_p dT$$

Chemical potential + thermal energy

$$\left(\frac{\partial H}{\partial T} \right)_p = c_p$$

Mass transport

$$\frac{\partial \rho_\alpha}{\partial t} + \nabla \cdot (\rho_\alpha \mathbf{V}_\alpha) = \dot{\varpi}_\alpha$$



$$Y_\alpha = \rho_\alpha / \rho$$

$$\frac{\partial \rho Y_\alpha}{\partial t} + \nabla \cdot (\rho Y_\alpha \mathbf{V}_\alpha) = \dot{\varpi}_\alpha$$



$$\mathbf{v}_{d\alpha} = \mathbf{V}_\alpha - \mathbf{V} \quad ; \quad \mathbf{V} = \frac{\sum \rho \mathbf{V}_\alpha}{\rho}$$

$$\rho \frac{D Y_\alpha}{Dt} \equiv \rho \frac{\partial Y_\alpha}{\partial t} + \underbrace{\rho \mathbf{V} \cdot \nabla Y_\alpha}_{\text{convection}} = - \nabla \cdot (\rho Y_\alpha \mathbf{v}_{d\alpha}) + \dot{\varpi}_\alpha \quad \underbrace{- \nabla \cdot (\rho Y_\alpha \mathbf{v}_{d\alpha})}_{\text{diffusion}}$$



$$\rho Y_\alpha \mathbf{v}_{d\alpha} \approx -\rho D_\alpha \nabla Y_\alpha \quad \text{Fick's law}$$

$$\rho \frac{D Y_\alpha}{Dt} = \frac{\partial \rho Y_\alpha}{\partial t} + \nabla \cdot (\rho Y_\alpha \mathbf{V}) = \nabla \cdot (\rho D_\alpha \nabla Y_\alpha) + \dot{\varpi}_\alpha \quad D_\alpha = \frac{\mu}{Sc_\alpha} \quad Le_\alpha = \frac{Pr_\alpha}{Sc_\alpha}$$

$$\rho = \sum \rho_\alpha, \quad \sum \dot{\varpi}_\alpha = 0, \quad \sum Y_\alpha \mathbf{v}_{d\alpha} = 0 \quad \frac{D\rho}{Dt} \equiv \frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0$$

Mass transport

Distribution of mole fraction

$$\nabla X_\alpha = \sum_{\beta} \frac{X_\alpha Y_\beta}{D_{\alpha\beta}} (\mathbf{v}_\beta - \mathbf{v}_\alpha) + (Y_\alpha - X_\alpha) \frac{\nabla p}{p}$$

$$+ \frac{\rho}{p} \sum_{\beta} Y_\alpha Y_\beta (\mathbf{f}_\alpha - \mathbf{f}_\beta) + \sum_{\beta} \frac{X_\alpha Y_\beta}{\rho D_{\alpha\beta}} \left(\frac{D_{T\beta}}{Y_\beta} - \frac{D_{T\alpha}}{Y_\alpha} \right) \frac{\nabla T}{T}$$

In case of two species

$$Y_1 \mathbf{v}_1 = -D_{12} \left\{ \nabla Y_1 + \frac{Y_1 Y_2}{X_1 X_2} (Y_1 - X_1) \frac{\nabla p}{p} + \frac{(Y_1 Y_2)^2}{X_1 X_2} \frac{\rho}{p} (\mathbf{f}_1 - \mathbf{f}_2) + \frac{D_{T,1}}{\rho D_{T,2}} \frac{\nabla T}{T} \right\}$$

Fick's law

$$Y_\alpha Y_\beta (\mathbf{v}_\beta - \mathbf{v}_\alpha) = Y_\alpha Y_\beta (\mathbf{V}_{d\beta} - \mathbf{V}_{d\alpha}) = Y_\alpha \underline{(Y_\beta \mathbf{V}_{d\beta})} - Y_\beta \underline{(Y_\alpha \mathbf{V}_{d\alpha})} = \underline{(Y_\alpha + Y_\beta)} (-Y_\alpha \mathbf{V}_{d\alpha}) = 1$$

Energy transport

Theory.

$$\rho dh = dp + \rho T ds$$

Enthalpy eq.

$$\rho \frac{Dh}{Dt} = \frac{Dp}{Dt} + \nabla \cdot \lambda \nabla h - \sum \nabla \cdot \rho D \sum (1 - Le_\alpha) \nabla Y_\alpha + \Phi + Q_R$$

Temperature eq.

$$\rho C_p \frac{DT}{Dt} = \frac{Dp}{Dt} + \nabla \cdot \lambda \nabla T + \sum c_{p,\alpha} \mathbf{j}_\alpha \cdot \nabla T + \Phi + Q_R + \dot{Q}_{reac}$$

$$dh = \sum Y_\alpha h_\alpha = \sum (h_\alpha^\circ dY_\alpha + c_{p,\alpha} Y_\alpha dT), \quad h_\alpha = h_\alpha^\circ + \int_{T_0}^T c_{p,\alpha} dT$$

Reactive heat

$$\sum h_\alpha^\circ \dot{\varpi} = -\dot{Q}_{reac.} : \text{Reactive heat}$$

$$C_p = \sum Y_\alpha C_{p,\alpha} \quad \text{Specific heat}$$

$$\text{Heat flux (Fourier's law)} \quad \mathbf{q} = -\lambda \nabla T$$

$$\alpha = \lambda / \rho C_p \quad \text{Thermal conductivity}$$

$$\text{Mass fux (Fick's law)} \quad \mathbf{j}_\alpha = -\rho D_\alpha \nabla Y_\alpha$$

$$Le_\alpha = \lambda / (\rho C_p D_\alpha) = \alpha / D_\alpha$$

$$\text{Low Ma No. apprrox.} \quad \nabla p \approx 0 \quad \therefore \frac{Dp}{Dt} \equiv \frac{\partial p}{\partial t} + \mathbf{v} \cdot \nabla p \approx \frac{dp}{dt}$$

Solution of combustion flow

convection	interaction	diffusion	external-force
$\frac{\partial \rho}{\partial t} = -u_j \frac{\partial \rho}{\partial x_j}$	$-\rho \frac{\partial u_k}{\partial x_k}$ $-\frac{\partial p}{\partial x_i}$	$+M$ $+ \frac{\partial}{\partial x_j} \left(\frac{\mu}{3} I \delta_{ij} + 2\mu S_{ij} \right) + F_i$ $+ \frac{\partial}{\partial x_j} \left(k \frac{\partial T}{\partial x_j} \right) + \Phi + Q$	mass
$\rho \frac{\partial u_i}{\partial t} = -\rho u_j \frac{\partial u_i}{\partial x_j}$	$-p \frac{\partial u_k}{\partial x_k}$		momentum
$\rho c_v \frac{\partial T}{\partial t} = -\rho c_v u_j \frac{\partial T}{\partial x_j}$	$p = \rho R T$		energy
 Heat source  Weak relation by low Ma approx.			
Additional eqs.		$+ \frac{\partial}{\partial x_j} \left(D \frac{\partial Y_\alpha}{\partial x_j} \right) \dot{\omega} +$	chemical spices
$\rho \frac{\partial Y_\alpha}{\partial t} = -\rho u_j \frac{\partial Y_\alpha}{\partial x_j}$			

Exercise 2c

Consider a system of equations for complex flow phenomena and analyze its coupling process to fluid dynamics;

- Ex. Multi-phase flow
- Buoyancy flow
- Magneto-hydrodynamics
- Flow in porous media
- Plasma flow etc.